Milestone Review Flysheet

nstitution pring Grove Area High School (Team Darwir

Milestone

CDR

Vehicle Properties		
Total Length (in)	88.125	
Diameter (in)	3.9	
Gross Lift Off Weigh (lb)	17.3	
Airframe Material	Fiberglass	
Fin Material	G10 Fiberglass	
Drag	0.95	

Stability Analysis			
Center of Pressure (in from nose)	64.2		
Center of Gravity (in from nose)	51.1		
Static Stability Margin	3.3		
itatic Stability Margin (off launch rail			
Thrust-to-Weight Ratio	17.3		
Rail Size and Length (in)	96		
Rail Exit Velocity(ft/s)	84.8		

	Recovery System Properties			
	Dogue Parachute			
Manufactu	irer/Model	Fr	uitychutes/I	FC
Si	ze		24 in	
Altitude	e at Deployn	nent (ft)	5184.58	
Velocity	at Deploym	ent (ft/s)	4.05	
Terminal Velocity (ft/s)		r (ft/s)	72.7	
Recovery Harness Material		Material	Tubular Nylon	
Harness Size/Thickness (in)		1		
Recovery Harness Length (ft)		ength (ft)	15	
Harness/Airframe structural		s will be attached to key components via quick ly secure the harness to the rocket		
Enerfy of	Section 1	Section 2	Section 3	Section 4
Each Section (Ft-	157	437.7	755.6	

Recovery Electonics		
Altimeter(s)/Timer(s) (Make/Model)	PerfectFlite CF Altimeters	
Redundancy Plan	The E-Bay will have 2 altimeters, each altimeter will have 2 charges, one for drouge and the other main. 2 altimeters will give it a redundant system.	

Motor Properties		
Motor Manufacturer	Cesaroni	
Motor Designation	K1200	
Max/Average Thrust (Ib)	306.5/268.3	
Total Impulse (lbf-s)	452.6	
Mass Before/After Burn	57.56oz/22.95oz	
Liftoff Thrust (lb)	306.5	

Ascent Analysis		
Maximum Veloxity (ft/s)	703.4	
Maximum Mach Number	0.625	
Maximum Acceleration (ft/s^2)	474.7	
Target Apogee (From Simulations)	5422	
Stable Velocity (ft/s)	44	
Distance to Stable Velocity (ft)	27	

Recovery System Properties				
Main Parachute				
Manufactu	irer/Model		Fruitychutes	5
Si	ze		72 in	
Altitude	e at Deployn	nent (ft)	600	
Velocity	at Deploym	ent (ft/s)	72	2.7
Term	inal Velocity	r (ft/s)	20.1	
Recovery Harness Material		Material	Tubular Nylon	
Harness Size/Thickness (in)		ness (in)	1	
Recovery Harness Length (ft		ength (ft)	25	
Harness/Airframe structural		s will be attached to key I components via quick ely secure the harness to the rocket		
Enerfy of	Soction 1		Section 3	Section 4
Each Section (Ft-	12	33.5	57.8	

Recovery Electonics		
Rocket Locators (Make/Model)	Communications Specialists Inc. R- 300 R/C ELT Receiver	
Transmitting Frequencies	***Required by CDR***	
Black Powder Mass Drogue Chute (grams)	3	

Pad Stay Time (Launch Configuration)	Each altimeter will have a battery life of 4 hrs.	Black Powder Mass Main Chute (grams)	3

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	Autonomous Ground Support Equipment (MAV Teams Only)
	Overview
Capture Mechanis m	
	N/A
	Overview
Container Mechanis m	
	N/A
Launch	Overview
Rail Mechanis m	N/A
Igniter	Overview
Installation Mechanis	
m	N/A

	Payload
	Overview
Payload 1	The payload will test the effect that the rocket's flight and acceleration have on the planaria's ability to regenerate
	Overview
Payload 2	
	N/A

	Test Plans, Status, and Results
Ejection Charge Tests	Each ejection charge will be 3.0 g of black powder. These charges will eject both our main and drouge parachutes through the seperation of the launch vehicle. These ejections are triggered by altimeters.
Sub-scale Test Flights	Successfull subscale test flight November 21st,2015. This subscale rocket was built at a 60% scale of what the full scale will be. Two test flights were launched with the first reaching a height of 1970' and the second reaching apogee at 2104'. Each flight had a successful deployment of the main parachute at 700'.
Full-scale	

Full scale test and ground ejection charge test scheduled for the weekend of January 15-16.

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Additional Comments

-Section 1(Nose Cone) Section 2(E-Bay and front body tube section) -Section 3(Rear body tube section including fins and motor casing)