Milestone Review Flysheet

Institution Spring Grove Area High School

Milestone	PDR
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Ve	ehicle Properties
Total Length (in)	85.25
Diameter (in)	4
Gross Lift Off Weigh (lb)	20.61
Airframe Material	Fiberglass
Fin Material	G10 Fiberglass
Drag	Cd. 75

Stability Anal	ysis
Center of Pressure (in from nose)	55.48 in
Center of Gravity (in from nose)	52.34
Static Stability Margin	2.14
Static Stability Margin (off launch rail)	2.38
Thrust-to-Weight Ratio	5.81
Rail Size and Length (in)	72
Rail Exit Velocity	43.23

Recovery System Properties					
	Dogue Parachute				
Manufactu	ırer/Model	F	RUITYCHUTES/IFC		
Si	ze		24 in		
Altitu	ide at Deployme	nt (ft)	5326.97		
Veloci	ty at Deploymer	nt (ft/s)	1.	11	
Terminal Velocity (ft/s)		73.7638			
Recovery Harness Material		Tubular Nylon			
Harness Size/Thickness (in)		1			
Recovery Harness Length (ft)		gth (ft)	15		
The harness is going be attatched to key structural components via quick links to safe secure the harness to the rocket.		ck links to safely			
Kinetic Enerfy	Section 1	Section 2	Section 3	Section 4	
of Each Section (Ft-Ibs)	82242.2199	17877.2722	18601.7544	N/A	

Rec	covery Electonics
Altimeter(s)/Timer(s) (Make/Model)	Perfectflite CF Altimeters
Redundancy Plan	The ebay will have 2 altimeters. Each altimeter will have 2 charges located on either side of the e-bay, one for drogue and the other for the main. Two altimeters will give it a redundant system.
Pad Stay Time (Launch Configuration)	Each altimeter will have a battery life of 4 hours.

Motor Properties	
Motor Manufacturer	Cesaroni
Motor Designation	K570
Max/Average Thrust (lb)	201/143
Total Impulse (lbf-s)	2062.9
Mass Before/After Burn	3.715 lb/ 1.532 lb
Liftoff Thrust (lb)	201

Ascent Analys	is	
Maximum Veloxity (ft/s)	605.49	
Maximum Mach Number	0.538	
Maximum Acceleration (ft/s^2)	14.36	
Target Apogee (From Simulations)	5326.97	
Stable Velocity (ft/s)	43.9993	
Distance to Stable Velocity (ft)	3.745	

Recovery System Properties					
Main Parachute					
Manufactu	ırer/Model	F	RUITYCHUTES/I	A	
Si	ze		72 in		
Altitu	de at Deployme	ent (ft)	600		
Veloci	ty at Deploymer	nt (ft/s)	73.7	7638	
Terminal Velocity (ft/s)		ft/s)	16.249		
Recovery Harness Material		Tubular Nylon			
Harness Size/Thickness (in)		1			
Recovery Harness Length (ft)		gth (ft)	25		
The harness is going be attatched to ke structural components via quick links to sa secure the harness to the rocket.		k links to safely			
Kinetic Enerfy	Section 1	Section 2	Section 3	Section 4	
of Each Section (Ft-lbs)	399.0831	867.497	902.6526	N/A	

Recovery Electonics	
Rocket Locators (Make/Model)	Communications Specialists, Inc.
Nocket Locators (Wake/Woder)	R-300 R/C ELT Receiver
Transmitting Frequencies	***Required by CDR***
Black Powder Mass Drogue Chute (grams)	3.5
Black Powder Mass Main Chute (grams)	3.5

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	Autonomous Ground Support Equipment (MAV Teams Only)
	Overview
Capture Mechanism	N/A
	Overview
Container Mechanism	N/A
	Overview
Launch Rail Mechanism	N/A
	Overview
Igniter Installation Mechanism	N/A
	Payload Paylo
	Overview
Payload 1	Our payload is designed to test the rate at which air intake, through a turbine, will generate current on ascent.
	Overview
Payload 2	N/A
	Test Plans, Status, and Results
Ejection Charge Tests	Each ejection charge will be 3.5g of black powder. These charges will eject both our main and drogue parachutes through the seperation of the launch vehicle. These ejections are triggered by altimeters.
Sub-scale Test Flights	Subscale flight is scheduled for November of 2015.
Full-scale Test Flights	N/A

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Additional Comments

- The hole in the nose cone for the payload will create additional drag and, as a result, we may need to increase the motor size to meet the mile height.
 - Section 1 refers to the nose cone and payload.
 - Section 2 refers to the front body tube and e-bay.
 - Section 3 refers to the back body tube, fins, and motor casing.