Milestone Review Flysheet

Institutio	Spring G	rove Area	High Sch	ool (Team	Darwin)	Miles	tone		FRR		
	Veh	icle Prope	rties			Motor Properties				ŧ	
Total Le	ngth (in)		88.125			Motor Manufacturer		Cesaroni			
Diame	ter (in)		3.9			Motor De	signation		K650		
Gross Lift Of	ff Weigh (lb)		18.4			Max/Averag	e Thrust (lb)	1	163.96/147.6	0	
Airframe	Material		Fiberglass			Total Impi			393.64		
Fin Ma	aterial	(310 Fiberglas	S		Mass Before	e/After Burn	69	9.65/23.04 (o	z)	
Dr	ag		0.95			Liftoff Th	rust (lb)		163.96		
	Sta	bility Anal	ysis				As	cent Analy	rsis		
Center of	Pressure (in f	rom nose)	64	.24		Maxir	num Veloxity	/ (ft/s)	635.3		
Center of	Gravity (in fr	om nose)	52	.25		Maxin	num Mach Ni	umber	0.565		
Stati	c Stability M	argin	3	3		Maximun	n Acceleratio	n (ft/s^2)	363		
Static Stabili	ty Margin (of	f launch rail)				Target Apo	gee (From Si	mulations)	5444.32		
Thrus	st-to-Weight	Ratio	8	.6		Stable Velocity (ft/s)			44		
Rail S	ize and Leng	th (in)	1.0	96		Distance	to Stable Ve	locity (ft)	27		
Rail	Rail Exit Velocity(ft/s)		60	0.6							
	Recovery	System P	roperties				Recovery	/ System P	roperties		
		gue Paracl				Main Parachute					
Manufacturer/Model Fruitychutes/IFC						Manufacturer/Model Fruitychutes					
Siz	Size 24 in			Siz	ze		72 in				
Altitude at Deployment (ft)		5444.32			Altitude at Deployment (ft)		60	00			
Velocity at Deployment (ft/s)		1.08			Velocity	at Deployme	ent (ft/s)	56	.86		
Term	inal Velocity	(ft/s)	58	.86		Term	inal Velocity	(ft/s)	12	.56	
Recove	ry Harness N	1aterial	Tubular Nylon			Recove	ry Harness M	1aterial	Tubula	r Nylon	
Harnes	s Size/Thickn	ess (in)	1			Harness Size/Thickness (in)		ess (in)	1		
Recover	y Harness Le	ngth (ft)	1	5		Recover	y Harness Le	ngth (ft)	25		
The harnes Harness/Airframe structural co			ss will be attached to key omponents via quick links ecure the harness to the rocket			Harness/Airframe Interfaces		The harness will be attached structural components via qu to safely secure the harness rocket		a quick links	
Kinetic Enerfy of Each Section	Section 1	Section 2	Section 3	Section 4		Kinetic Enerfy of Each Section	Section 1	Section 2	Section 3	Section 4	
(Ft-lbs)	313	3,3	2310			(Ft-lbs)		, , , , , , , , , , , , , , , , , , ,	110		
	Reco	very Elect	onics				Reco	very Elect	onics		
Altimeter(s)/Timer(s) (Make/Model) PerfectFlite		lite CF Alt	lite CF Altimeters		Rocket Locators (Make/Model)		Communications Specialists Inc. R-300 R/C ELT Receiver				
Redundancy Plan altimeter w		altimeter wi	Bay will have 2 altimeters, each			Transmitting Frequencies		***Required by CDR***			
Pad Stay Time (Launch		drouge and the other main. 2 altimeters will give it a redundant system. Each altimeter will have a				Black Powder Mass Drogue Chute (grams)			2		
	122411011	Each alt	inieter Wi	ii flave a					3		

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Institutio	Spring G	rove Are	a High S	Schoo	ol (Team	n Darwin)	Mile	stone		FRR			
				- 6	I Cl			•• · · · · ·	- L\				
		Aut	onomou	us Gro	una Sup		pment (IVI) rview	AV Teams	Only)				
Capture			-					·					
Mechanism													
			. 			N	I/A						
		•				Ove	rview		1				
Container Mechanism													
						N	I/A						
-						Ove	rview						
Launch Rail Mechanism							I/A						
						18	/A						
						Ove	rview						
Igniter Installation								-					
Mechanism													
						N	I/A						
						Payload							
							rview						
Payload 1						-	<u></u>						
	The p	ayload will	test the e	ffect tha	at the rock	ket's flight an	d acceleratio	on have on the	e planaria's al	oility to regen	erate		
						Uve	rview						
Payload 2													
						N	I/A						
				Te	est Plans	s, Status, a	nd Results	S _.					
Ejection													
Charge Tests													
	Each ejection charge will be 3.0 g of black powder. These charges will eject both our main and drouge parachutes t seperation of the launch vehicle. These ejections are triggered by altimeters.			through the									
Sub-scale Test Flights	Successfull	Successfull subscale test flight November 21st,2015. This subscale rocket was built at a 60% scale of what the full scale will be.											
	Two test flig	hts were lau	inched wi	th the fi	irst reachi	ing a height o	of 1970' and t	as built at a 6 he second rea rachute at 70	aching apoge	e at 2104'. Ea	ach flight had		
		-	-		20000141	<u> </u>	the man pu	i dell'acc ac 70	Ÿ		-		
Full-scale													
Test Flights	Successful for 4585 ft v	ull scale fligh with a 4 grai	nts on Ma n K1200.	On Mar	rch 13th oi	ur rocket flev	v with our ne	n successful we we chosen moght of 4981 fe	tor annd flew	nunch flying to successfully	o a height of and was		

			Mi	ilestone	Review	v Flyshe	eet			
Institutio	Spring G	rove Area	High Sch	ool (Team	Darwin)	Miles	tone		FRR	
				Additi	onal Comr	ments				
		•	•						·	
-Sectio	on 1(Nos	e Cone) S	Section 2(section	E-Bay and on includin	front boo	dy tube se d motor ca	ection) asing)	-Section 3(I	Rear body	tube